



Load Planning in Modern Aviation – Computational Analysis

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ABSTRACT:

Load control plays a pivotal role in ensuring aircraft safety and efficiency. This paper explores real-world aircraft load planning, highlighting the balance between traditional manual methods and modern automation. Drawing from practical experience as a Load Officer at IndiGo, it examines the manual preparation of load sheets—including Zero Fuel Weight, fuel, and passenger distribution—and contrasts it with digital tools like IndiGo's Load and Trim Beta system. It emphasizes the continued relevance of manual cross-verification in the digital age, especially during last-minute operational changes. The study also sheds light on challenges in Centralized Load Control (CLC) and the evolving human-machine collaboration in aviation operations.

Keywords: Load and Trim, Centre of Gravity (CG), Aircraft Load Sheet, Load Control, Aviation Software, CLC

Introduction:

What is Centralized Load Control in Aviation?

Aircraft Load Control is one of the most vital yet often overlooked operations in aviation. It ensures that each flight remains within its permissible weight and balance limits—safeguarding flight stability, structural integrity, and fuel efficiency. This process is critical to the safety and performance of commercial aircraft such as the Airbus A320, A321, and ATR 72. As a Load Officer at IndiGo, the author draws from hands-on experience to analyse how accurate load planning directly influences real-world flight operations.

Historically, the practice of load control evolved from manual estimation techniques in the early 20th century to sophisticated computerized systems used in today's high-capacity airline environments. Load control began with paper-based load manifests in the 1920s and later transitioned through milestones such as electronic load sheets in the 1970s and fully automated Centralized Load Control (CLC) systems in the 21st century.

Load planning involves the systematic allocation of payload—passengers, cargo, baggage, and fuel—across the aircraft. Key calculations include Zero Fuel Weight (ZFW), Take off Weight (TOW), Landing Weight (LW), and precise Centre of Gravity (CG) placement, which are vital for safe aircraft performance across all flight phases. While manual calculations sharpen operational skills, they are increasingly being supplemented by advanced digital systems such as IndiGo's Load and Trim Beta software.

Inaccurate load planning can result in severe consequences such as unstable aircraft behaviour, excessive wear on airframes, or even regulatory violations. Hence, each commercial flight must be supported by an authorized Load and Trim Sheet validated by a certified Load Controller and often reviewed by the aircraft's Commander.

This paper investigates the significance of accurate load control, drawing from both qualitative and quantitative insights. It contrasts traditional manual load sheet preparation with modern automated methods, focusing on real-time load control practices for Airbus A320, A321, and ATR 72 aircraft under IndiGo's operational framework. Through this study, the paper aims to highlight how technology and human expertise intersect to ensure efficient and safe aircraft operations

What is the use of computational approach to load planning?

Manual and computational methods of load control represent two distinct phases in the evolution of airline ground operations. The manual approach, still taught as foundational training, requires the Load Controller to calculate weights, moments, and CG positions using zone-based distribution and trim charts. It demands high attention to detail, deep understanding of aircraft balance theory, and the ability to visualize the weight distribution effect on flight characteristics. Although accurate when executed correctly, it is time-consuming and vulnerable to human error, especially under time pressure or during last-minute changes.

In contrast, computational load planning, as seen in systems like IndiGo's **Load and Trim BETA**, is built for speed, reliability, and operational scalability. The software automates most calculations, drawing from dynamic data inputs like passenger counts, baggage scans, and cargo weights. It instantly determines CG positions and generates trim indexes with high accuracy. The software's logic includes safety envelopes, auto-balancing features, and real-time synchronization with fuelling and dispatch systems, which makes it highly efficient for high-frequency airline operations.

Methodology:

This study adopts a **mixed-method research approach**—incorporating both qualitative and quantitative data—to explore and evaluate modern load planning methods in aviation, with a specific focus on the operations of IndiGo Airlines and its Load and Trim BETA system.

Research Design

The research is designed as a **comparative case study**, examining two core load control approaches:

- **Manual Load sheet Preparation** (traditional method)
- **Computational Load Planning** using Load and Trim BETA (modern method)

These two models are analysed in terms of **accuracy, efficiency, safety compliance, and operational reliability**, particularly across Airbus A320, A321, and ATR 72 aircraft types.

Data Collection Methods

Primary Data:

- Observational insights and hands-on experience from the researcher's role as a Load Officer at IndiGo.
- Live examples and calculations drawn from actual load sheets generated during routine operations.

Secondary Data:

- Technical manuals such as the Airbus AOM and DGCA CAR regulations.
- Standard operating procedures (SOPs) and internal IndiGo documentation on load control.
- Industry reports and publications (e.g., IATA WBM, FAA Advisory Circulars).

Objective:

1. To analyse and compare manual and computational methods of aircraft load planning in terms of safety, efficiency, and operational accuracy.
2. To evaluate the effectiveness of IndiGo's Load and Trim BETA system in automating and optimizing weight and balance procedures for A320, A321, and ATR 72 aircraft.
3. To identify the practical challenges and limitations of both traditional and modern load control methods in real-time flight operations.

Results

Manual load sheet preparation is the traditional method of computing an aircraft's weight and balance without relying on automation. This process relies heavily on human calculation and requires a strong grasp of aircraft configuration, standard weights, and the centre of gravity (CG) concepts. In this chapter, we break down each component of a **manual load sheet for an A321-251NX/A321-252NX** aircraft using a DGCA-approved sample, illustrating how each figure is derived and how this affects flight safety.

Load and Trim Calculation Process in A321-251NX/A321-252NX Aircraft

In commercial aviation, load and trim calculations are essential to ensure that aircraft operate within safe structural and aerodynamic limits. The Airbus A321-251NX and A321-252NX, configured in a dual-class seating layout, are subject to specific weight limitations defined by the manufacturer and regulatory authorities. These include a Maximum Zero Fuel Weight (MZFW) of approximately 75,500 kg, a Maximum Take off Weight (MTOW) of 89,000 kg, and a Maximum Landing Weight (MLW) of 75,000 kg. The aircraft is typically divided into several passenger zones (Z1, Z2, Z3, etc.) and cargo compartments (Forward, Aft, and Bulk), each contributing to the aircraft's centre of gravity (CG) through individual calculated moments.

The process begins with identifying the Basic Empty Weight (BEW), which includes the structural weight of the aircraft along with permanently installed equipment. The Dry Operating Weight (DOW) is then calculated by adding the crew, pantry supplies, and other fixed items to the BEW. This DOW, along with its corresponding moment (force \times distance), serves as the foundation for subsequent load computations. The payload, which includes passengers, baggage, and cargo, is then determined. Passenger weights are calculated by multiplying the number of occupants in each zone with a standard weight per passenger, generally 75 kg as per DGCA or airline policy. Baggage is estimated based on an average weight per bag—typically ranging from 15 to 20 kg—and is distributed across different cargo holds. Additional cargo, listed separately on the load sheet, is added to complete the total payload.

Once the payload is determined, the Zero Fuel Weight (ZFW) is calculated as the sum of the DOW and payload. For centre of gravity computation, each component's moment is calculated by multiplying its weight with its respective arm (the distance from a defined reference point or datum). Summing all individual moments gives the total moment, and dividing this by the ZFW yields the CG position in terms of percentage of the Mean Aerodynamic Chord (% MAC). This CG value is then converted into a Trim Index using aircraft-specific trim charts.

IndiGo LOADSHEET AND LOADMESSAGE
PASSENGER AIRCRAFT (ALL WEIGHTS IN KILOGRAM)

Address(es) _____

Originator ☐ LDM ☐ Flight 4E / ☐ AC Reg ☐ Version ☐ J12/208Y ☐ Crew ☐ BA ☐ Date _____

OPERATING EMPTY WEIGHT
Crew _____
Party _____

MAXIMUM WEIGHT FOR
Take-off Fuel ☐ 7 5 6 0 0

ZERO FUEL
7 5 6 0 0

TAKE-OFF
7 9 2 0 0

LANDING
7 9 2 0 0

TRIP FUEL ☐ 0

ALLOWED WEIGHT FOR
TAKE-OFF (Lowest of a, b or c) _____

Operating Weight ☐ 0

ALLOWED TRAFFIC LOAD _____

OPERATING WEIGHT _____

IndiGo

LOAD IN LOWER COMPARTMENT

Weight	Index	Weight	Index	Weight	Index
40.00	1.00	40.00	1.00	40.00	1.00
41.00	1.01	41.00	1.01	41.00	1.01
42.00	1.02	42.00	1.02	42.00	1.02
43.00	1.03	43.00	1.03	43.00	1.03
44.00	1.04	44.00	1.04	44.00	1.04
45.00	1.05	45.00	1.05	45.00	1.05
46.00	1.06	46.00	1.06	46.00	1.06
47.00	1.07	47.00	1.07	47.00	1.07
48.00	1.08	48.00	1.08	48.00	1.08
49.00	1.09	49.00	1.09	49.00	1.09
50.00	1.10	50.00	1.10	50.00	1.10
51.00	1.11	51.00	1.11	51.00	1.11
52.00	1.12	52.00	1.12	52.00	1.12
53.00	1.13	53.00	1.13	53.00	1.13
54.00	1.14	54.00	1.14	54.00	1.14
55.00	1.15	55.00	1.15	55.00	1.15
56.00	1.16	56.00	1.16	56.00	1.16
57.00	1.17	57.00	1.17	57.00	1.17
58.00	1.18	58.00	1.18	58.00	1.18
59.00	1.19	59.00	1.19	59.00	1.19
60.00	1.20	60.00	1.20	60.00	1.20
61.00	1.21	61.00	1.21	61.00	1.21
62.00	1.22	62.00	1.22	62.00	1.22
63.00	1.23	63.00	1.23	63.00	1.23
64.00	1.24	64.00	1.24	64.00	1.24
65.00	1.25	65.00	1.25	65.00	1.25
66.00	1.26	66.00	1.26	66.00	1.26
67.00	1.27	67.00	1.27	67.00	1.27
68.00	1.28	68.00	1.28	68.00	1.28
69.00	1.29	69.00	1.29	69.00	1.29
70.00	1.30	70.00	1.30	70.00	1.30
71.00	1.31	71.00	1.31	71.00	1.31
72.00	1.32	72.00	1.32	72.00	1.32
73.00	1.33	73.00	1.33	73.00	1.33
74.00	1.34	74.00	1.34	74.00	1.34
75.00	1.35	75.00	1.35	75.00	1.35
76.00	1.36	76.00	1.36	76.00	1.36
77.00	1.37	77.00	1.37	77.00	1.37
78.00	1.38	78.00	1.38	78.00	1.38
79.00	1.39	79.00	1.39	79.00	1.39
80.00	1.40	80.00	1.40	80.00	1.40
81.00	1.41	81.00	1.41	81.00	1.41
82.00	1.42	82.00	1.42	82.00	1.42
83.00	1.43	83.00	1.43	83.00	1.43
84.00	1.44	84.00	1.44	84.00	1.44
85.00	1.45	85.00	1.45	85.00	1.45
86.00	1.46	86.00	1.46	86.00	1.46
87.00	1.47	87.00	1.47	87.00	1.47
88.00	1.48	88.00	1.48	88.00	1.48
89.00	1.49	89.00	1.49	89.00	1.49
90.00	1.50	90.00	1.50	90.00	1.50
91.00	1.51	91.00	1.51	91.00	1.51
92.00	1.52	92.00	1.52	92.00	1.52
93.00	1.53	93.00	1.53	93.00	1.53
94.00	1.54	94.00	1.54	94.00	1.54
95.00	1.55	95.00	1.55	95.00	1.55
96.00	1.56	96.00	1.56	96.00	1.56
97.00	1.57	97.00	1.57	97.00	1.57
98.00	1.58	98.			

Fig 1 Airbus A321 Manual Load sheet

For instance, using sample values: if the DOW is 48,000 kg with a moment of 1,200,000, the passenger load is 11,250 kg with a moment of 300,000, and the baggage/cargo load is 6,000 kg with a moment of 180,000, the ZFW would be 65,250 kg. The total moment would then be 1,680,000, giving a CG of approximately 25.75% MAC, which corresponds to a Trim Index of around +3 to +4 on the trim chart.

Fuel planning follows the payload computation. The take-off fuel (TOF) and trip fuel are factored into the calculation to determine the Ramp Weight (RW) and Take off Weight (TOW), with fuel distribution across different tanks also contributing to the overall aircraft moment. The TOW is computed by adding the ZFW and TOF, then subtracting the taxi fuel. The Landing Weight (LW) is subsequently calculated by deducting the trip fuel from the TOW. Each of these weights—ZFW, TOW, and LW—must be associated with their respective CG values, which must remain within the acceptable trim envelope as prescribed in the aircraft operations manual.

Finally, load sheet preparation culminates in the regulatory compliance phase. The completed document is signed by the Load Controller and the Pilot-in-Command (PIC). As per the Directorate General of Civil Aviation (DGCA) guidelines, load sheets must comply with CAR Section 8 Series 'S' Part I and ICAO Annex 6, and must adhere strictly to the airline's Standard Operating Procedures (SOPs). These load sheets are legally mandated to be preserved for audit and safety review purposes. Ensuring precise and compliant load sheet documentation is fundamental to safe and efficient aircraft operation.

Refer to Fig 2 for the landing page of application software Load and Trim Beta designed by IndiGo to carry out their Load Control Operations. Now we will analyse how computerised load control work

IndiGo's *Load and Trim BETA* system is a proprietary computerized load control platform designed to automate and streamline aircraft weight and balance operations. Integrated within the airline's ground handling and operations network, this software supports real-time computations of weight distribution, automatic zone allocation, and generation of DGCA-compliant load sheets for aircraft such as the A320, A321neo, and ATR 72-600. The system minimizes manual entry errors by importing data from multiple sources including the Departure Control System (DCS) for passenger and baggage details, cargo terminals for freight weights, and dispatch or fuelling agencies for fuel data. Once data is entered, the system automatically allocates payload across cabin zones and cargo compartments, optimizing the aircraft's centre of gravity (CG) based on configuration. Using embedded OEM trim charts, it calculates and validates the CG and Trim Index for Zero Fuel Weight (ZFW), Take-off Weight (TOW), and Landing Weight (LW), ensuring they fall within operational safety envelopes. The finalized load sheet, complete with required signatures, is transmitted electronically via ACARS or uploaded to the Electronic Flight Bag (EFB) for crew access. The platform also provides the stabilizer trim setting required by the cockpit, ensuring the Pilot-in-Command reviews and confirms the data before departure. Overall, *Load and Trim BETA* significantly enhances accuracy, compliance, and communication in load control operations.

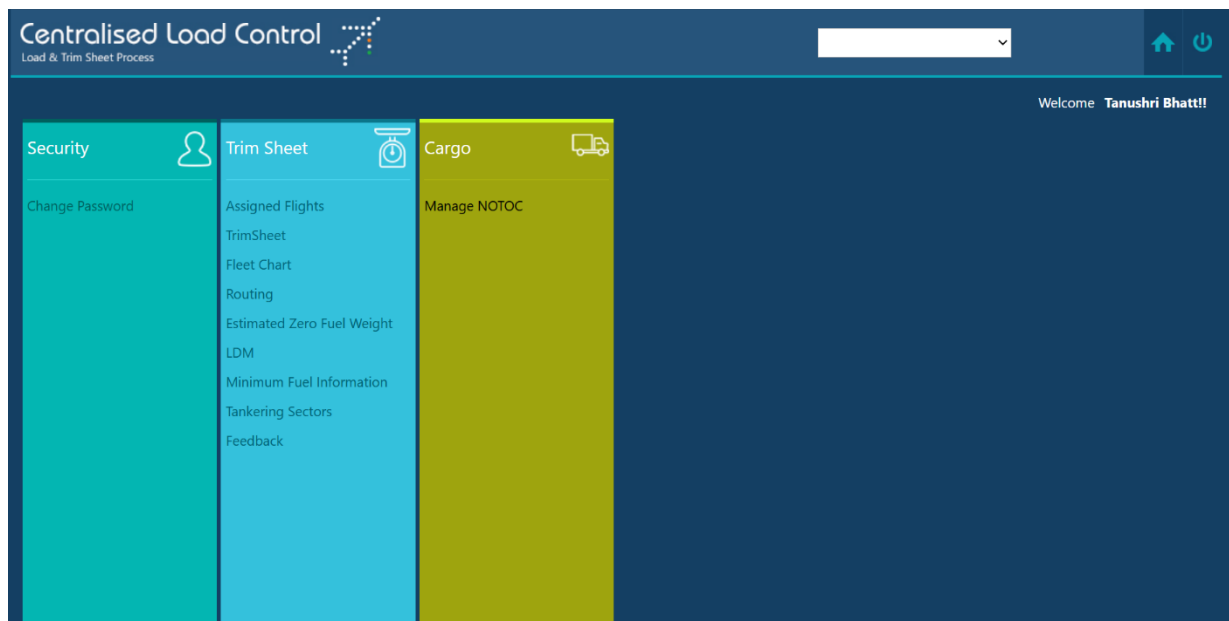


Fig 2 Load Control Software BETA

Conclusion

In today's precision-driven aviation environment, **load planning has evolved from a manual, chart-based procedure to a sophisticated, computation-led process**. This research has explored the critical role of load control in ensuring aircraft safety, fuel efficiency, and operational reliability by using a comparative study between manual and online generation of a load sheet.

Through a focused analysis of IndiGo's Load and Trim BETA system, the study demonstrates how computerized load planning enhances accuracy, reduces human error, and accelerates turnaround time. However, the research also highlights that manual methods continue to play a supporting role, especially during contingencies or technical downtimes, underscoring the need for hybrid proficiency.

A review of global leaders like Qatar Aviation Services reveals that while Indian carriers are progressing steadily, further integration, automation, and predictive analytics could close the operational gap. Ultimately, the future of load planning lies in **leveraging computational tools intelligently**—combining algorithmic precision with human oversight to achieve optimized, safe, and scalable ground operations in aviation.

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