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Design and Fabrication of Radio Controlled N-Chia Drone for Surveillance

¹Sneha Kashyap, ²Ritvik Jaiswal, ³Tushar Tyagi, ⁴Rakesh Kumar, ⁵Dhananjay Singh

1,2,3,4Student, 5Assistant Professor

¹Department of Electronics and Communication,

¹Noida Institute of Engineering and Technology, Greater Noida, India

ABSTRACT :

The unmanned aerial vehicle, or UAV, is the subject of this study effort, which focuses on modifications to the UAV's present design process. A person on the base can operate a radio-controlled plane, also known as an RC plane, using a handheld device. Controlling the transmitter allows you to command the receiver, which in turn sends commands to the aircraft's servos via the transmitter. The key information required to build a flying model has been attempted to be covered in this research endeavour. We think that helping students with this will help them finish their models. Designing, building electronics, and surveillance are the four main aspects of aeroplane modelling that this study primarily focuses on.

Keywords-RC plane, Unmanned Air Vehicle (UAV), Airfoil, Fuselage, surveillance

INTRODUCTION

In the modern world, planes are familiar to us because we have all seen or possibly even experienced one flying at some point in our lives. However, the concept of developing a technology to enable human flight had already started to take flight in human brains quite some time ago. Many well-known historical narratives feature the concept of flight, such as the Greek myth of Icarus and Daedalus and the well-known Indian fantastic tale of the Ramayana, which features a flying machine called "Vimana." The over-wing designs of birds were also studied by one of history's most brilliant minds, Leonardo da Vinci, who in 1502 devised a man-controlled aircraft. Many of these academics, including Lawrence Hargrave, Otto Lilienthal, and Sir George Cayley, made contributions to the field of streamlined features. The cost, weight, performance, and capacities of engines, batteries, and gadgets have all improved dramatically over the years, which has led to a huge growth in the hobby of flapping RC aeroplanes. There are numerous models and fashion trends accessible.

Basic Design Parameters

Wing loading

According to aerodynamics, the amount of load carried by each unit of a wing's area is referred to as the wing loading.

When a smaller wing is utilised in conjunction with a fuselage, the amount of weight carried by each unit wing area

increases instability. A larger wing reduces the load distribution on the wing-surface and allows for a considerably more

stable flight. For model aircraft, wing load is measured in ounces per square foot (oz/ft2).

Wing Loading (oz/sq.ft) = Weight (oz) / Wingarea(ft)

Dihedral Angle

The angle between the x-y axis and the centre of the wing is known as the dihedral angle. To achieve aircraft symmetry, the right and left parts of each wing must have the same dihedral angle. Dihedral is used to provide flight stability. The phrase "positive dihedral" or "dihedral" is used when the angle is greater than that of "x-y". The term "negative dihedral" or "anhedrals" (wing tip) is used when the angle is smaller than the angle of the "x-y" [wing].

Aspect Ratio

The wings' overall efficiency is demonstrated by their aspect ratio. The wings are measured for both length and slenderness. The "high" aspect ratio aircraft wing is long and narrow, whereas the "low" aspect ratio is broad and unyielding. In proportion to the increase in attack angle, the high aspect ratio wing creates higher elevation than the low aspect ratio wing.

Aspect ratio = wing $span^2 / wing$ area

Lift coefficient

According to the formula, the lift force depends on the air density r, airspeed V, the lift coefficient of the wing, and the area of the wing.

Lift Force = 0.5 * r * V 2 * Wing's Lift Coefficient * Wing's Area

Wing's Lift Coefficient

It depends on the kind of airfoil, the number of Reynolds (Re), the aspect Ratio (AR), and the Attack Angle (alpha) prior to the stall Angle. It is a dimensionless quantity. The formation of wing lift nevertheless results in driven and parasitic drag, which work against the passage of the aeroplane through the air. The air r, air velocity V, wing drag coefficient, and region-specific area of the wing's formula all influence the density of the produced drag:

Drag Force = 0.5 * r * V 2 * Wing's Drag Coefficient * Wing's Area

Wing's Drag Coefficient

It is a dimensionless feature that is influenced by the shape of the wings, aspect ratio, attack angle, and Reynolds number, much like lift coefficient. Graphs that show the lift-drag volume attained at various angles and the lift/drag ratio can be used to illustrate the properties of a specific airfoil. The lifting and drag coefficients of the same airfoil differ under various Reynold numbers.

LITERATURE REVIEW

In paper [1] authors proposed an overview of the communication system is imperfect without understanding the activity Unmanned aerial vehicles (UAVs) are becoming more and more common in a range of civil applications, such as real-time monitoring, wireless coverage, remote sensing, search and rescue, cargo delivery, security and surveillance, precision agriculture, and civil infrastructure inspection. A wide range of facets of human life are being significantly impacted by this technology. In order for the aerospace sector to grow and remain competitive, UAVs have become essential.

In paper [2] authors proposed a hand-held transmitter and a receiver installed inside the aircraft are used to remotely pilot a radio-controlled (model) aircraft, also known as an RC plane or aircraft. With more effective motors (both electric and small internal combustion and jet engines), lighter and more powerful batteries, and less expensive radio systems, the pastime of flying RC aircraft has gained popularity all over the world.

In paper [3] authors proposed this paper was carried out aiming the modification in the present design process for Unmanned Air Vehicles (UAV). UAVs are flying machines, that are either self-piloted or flown by a pilot. The desired goal was fulfilled by designing and fabricating a lightweight, twin boomed, U tail RC Plane. The design was originally modelled in Solid Works. The airfoil analysis was carried out in xflr5 v6 and then verified by ANSYS 2019 R2 (Academic Version). Wing analysis was done in ANSYS 2019 R3, for obtaining static and total pressure distribution. Finally, the model was analyzed in both aerial and ground tests.

per [4] authors proposed this paper focuses on developing an automatic flight control system for a fixed-wing unmanned aerial vehicle (UAV) using a software-in-the-loop method in which the PID controller is implemented in National Instruments LabVIEW software and the fixed-wing UAV flight dynamics are simulated using the X-Plane flight simulator. The Plane Maker software was used to generate the fixed-wing UAV model, which is based on existing geometry and propulsion data from the literature.

In paper [5] authors proposed this project involves developing a radio-controlled aircraft while taking into account different factors, like taper ratio, aspect ratio, and other factors. In solid edge, a 3D CAD model is created using the aforementioned considerations. Prior to the construction of the aircraft, the quantitative and qualitative characterization of the aircraft wing will provide useful information to verify the choice and design of the wing. With the aid of calculated values, the geometry of the wings is first constructed, and then the designed wings are analysed for various boundary conditions.

ELECTRICALLY POWERED RC AIRCRAFT

Wing Selection

All these birds alike robots are called "flying ornithopters." There are three types of wings: rectangular, elliptical, and tapered. The most commonly involved wings for the radio-controlled birds are rectangular. These types of wings are best for utilization, according to the manufacturer's perspective. These types of wings tend to slow down rest at the root wing and also give satisfactory slowdown advance notice, sufficient aileron adequacy, and are generally very steady. It is likely preferred for the minimum expenses and low speed of the radio-controlled birds.

Air foil selection

It is a shape of the wing, and it moves through a fluid that produces aerodynamic force. although the component of this force is perpendicular to the direction of the motion, which is known as lift. The lift is primarily the result of its angle of attack and shape. The below figure represents the air foil terminology, and by modifying all the above features of an air foil, it can also adapt the performance if the wing is suitable for the task.

Fuselage Selection

Fuselage design focused on three different models, namely the conventional monoplane, the biplane, and the n-plane. Subsequently, a traditional plan was viewed as frequently preferred inside the model structure local area because of its simplicity of development. Hence, in this paper, the regular sort of fuselage was utilized.

Tail Design or Empennage Design

The empennage, otherwise called the tail or tail assembly, of most aeroplanes gives strength to the airplane, likewise to the quills on a bolt. Despite successful control surfaces, numerous early aeroplanes that needed balancing out empennage were basically not flyable. Today, a couple of (frequently somewhat unsteady) heavier-than-air aeroplanes can fly without empennage.

TECHNICAL REQUIREMENTS

BLDC Motor

A brushless DC motor is the full form of a BLDC motor. Because no brush is used to transfer electricity between the rotor and stator in a BLDC motor, it is known as a brushless dc motor. Brushless direct current (BLDC) motors have high efficiency and excellent controllability.

ESC

The phrase ESC stands for "Electronic Speed Control" is an electronic device designed for controlling the speed of an electric motor.

Lipo Battery

The lithium-ion polymer battery is a revived lithium-ion battery that employs a polymer electrolyte rather than a liquid electrolyte. They have a much higher specific energy density than other types of lithium batteries and are used in applications where mass is critical, such as remote-control planes and some electric vehicles.

Servo Motor

A servo motor is a tiny motor that moves the control surfaces of RC planes. Little brush motors, a potentiometer, and a few gears are the components that make up servos. The potentiometer regulates the motor's motion, enabling the servo to move precisely the right amount of grease. Servos can rotate mechanically 180° , but they can only move 90° on their own.

Transmitter and receiver

A 2.4 GHz transmitter and 6-channel receiver were used to control elevator movement during takeoff and docking, as well as the motor's throttle. Each appendage—counselor, elevator, and motor—had its own channel. Two additional channels were left open.

Live Video Reception

A miniature camera viable with the regulator has been mounted the RC bird to give live observation. The live criticism from the RC bird will be gotten offboard on a PC.

HARDWARE CONNECTION

Hardware makeup for the controlling and other parts of the model's behaviour allow the operator to control its movements. This involves all the gadget parts, including the BLDC motor, speed controller, servo motor, battery, and other parts.

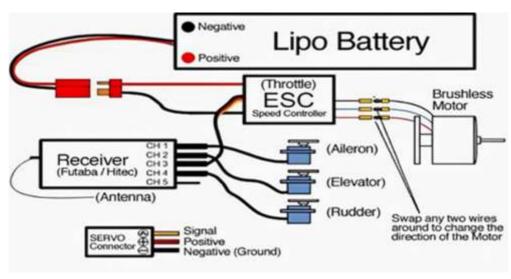


Fig1: Hardware connections on the RC bird plane

METHODLOGY

The project's methodology is as follows: 1. Determine the configuration type. 2. Parameter escalation to achieve the best staging 3. Adjusting the CG on the wing and chasis (for example, the neutral point on the wing). 4. Creating the core, wing, elevator, and so on (using the appropriate airfoil). 5. Examine the structure in terms of its criteria. 6. Analyze and disseminate information as needed.

FABRICATION

Based upon the above hardware connections, wing section, fuselage section, tail section was fabricated using foam material as shown in figure below.



Fig2: wing construction

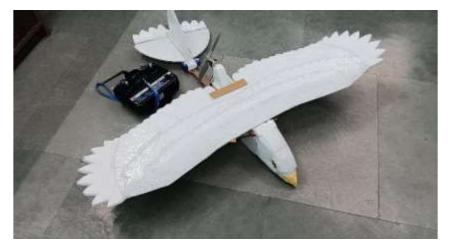


Fig3: Proposed prototype

CONCLUSION

An effort has been made to methodically design, fabricate, inspect, build, and fly a specimen radio-controlled N-Chia drone. An effort has been put together to methodically design, examine, and construct an airlift RC bird plane prototype. Through this project, the members of the group gained an understanding of the fundamentals of aircraft design, construction, engineering, and evaluation. The project also learns the basics of aircraft technology and utilises analysis and design tools.

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